# ACOUSTIC EMISSION FOR IDENTIFICATION OF THE DOMINANT STRESS COMPONENT IN POLYMER COMPOSITES AT EARLY LOADS

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# PRESENTATION OUTLINE

- Introduction
- Theoretical background
- Aim of the study
- Material and testing equipment
- Experimental results: quasi-static tests
- Experimental results: incremental loading tests
- Conclusions



## INTRODUCTION



- Reduction of weight → primary target in many engineering fields
- Fibre reinforced polymer composite materials → promising for reducing weight and CO<sub>2</sub> emissions
- Main advantage: <u>lightweight</u> <u>materials</u>





 Carbon Fibre Reinforced Polymer (CFRP): extra advantages like exceptional durability, application flexibility, corrosion resistance





- Composites: anisotropic materials
- Multiaxial stresses occur in the composite laminas even under uniaxial loading due to different fibre orientations (internal multiaxiality)





- **Damage sequence in composites** is complicated (interfacial debondings, matrix cracks, delaminations, fibre breaks)
- Even more complicated when multiaxial stresses occur → can lead to different mechanical response, influencing the structural integrity of the laminate







- Multiaxiality not extensively studied in literature
- Monitoring of damage with respect to different multiaxial stresses in lab conditions necessary → predictive tool for real applications
- Prediction of stress states and identification of dominant stresses essential even from early loading stages
- Acoustic Emission (AE) CAN be used to give solutions to these problems!





- Acoustic Emission: characterisation of damage of materials by interpreting the generated elastic waves
- Commonly applied in composite materials for investigations in the time domain and frequency domain
- Clustering approaches have been proposed
- No link to multiaxial stress states





- Feature analysis: selection of the most appropriate signal features
- It has been used so far for damage mode classification
- Rise Time (RT) and Average Frequency (AF) among the most popular





- Kaiser effect: the absence of detectable AE until the previously maximum applied stress is exceeded
- Felicity effect: the presence of detectable AE at stress levels below those previously applied → described by the Felicity Ratio (FR)
- $FR = \frac{\text{stress level at which AE resumes during a loading step}}{\text{maximum stress applied at the previous loading step}}$
- The Calm Ratio (CR) can be another damage parameter
  - $CR = \frac{AE \text{ activity during the unloading part of the cycle}}{AE \text{ activity over the total cycle}}$



# AIM OF THE STUDY

- To verify that AE can distinguish the different damage modes under multiaxial stress states
- Can AE indicate the dominant stress/strain component within the composite laminate from early loading stages?
- Which AE parameters are the most effective for such stress indications?



# MATERIAL AND TESTING EQUIPMENT

- To introduce different multiaxial stress states → two angle-ply carbon/epoxy laminates were tested
- Based on the multiaxiality ratio  $\lambda_{12} = \sigma_6 / \sigma_2$

 $\begin{array}{l} [0^{\circ}/30^{\circ}]_{2s}: \lambda_{12} = 2.02 \\ [0^{\circ}/60^{\circ}]_{2s}: \lambda_{12} = 0.64 \end{array}$ 

 $[0^{\circ}/30^{\circ}]_{2s}$ : dominant shear stresses  $[0^{\circ}/60^{\circ}]_{2s}$ : dominant transverse stresses





## MATERIAL AND TESTING EQUIPMENT



- Continuous static tests and interrupted tests displacement controlled at 1 mm/min rate
- Two Pico sensors for the AE acquisition → 35 dB threshold
- Digital Image Correlation (DIC) for strain measurements
- Through-the-thickness free-edge damage monitoring at regular steps with optical microscopy









 $85\% \sigma_{ult}$ 

00

30°

0°

30°

30°

30°

00

00



# $[0^{\circ}/60^{\circ}]_{2s} \text{ laminates}_{65\% \sigma_{ult}}$









































**MECHANICS OF MATERIALS** & CONSTRUCTIONS 

BRUSSEL













## CONCLUSIONS

- AE can be effectively used for the identification of damage in polymer composites
- Significant differences from early loading stages allowing <u>indications of the dominant stress</u> <u>component</u>
- RT good indicator for the identification of damage modes and the transition between modes
- Low RT linked to tensile related phenomena and high RT to shear related phenomena
- Continuous increase of RT when shear is dominant
- FR is characterised by reduction when delaminations occur → can be used as <u>damage mode</u> <u>transition indicator</u>
- FR appears lower values when shear is dominant → FR is not only material dependent, but also stress state dependent → can be used as <u>stress state indicator</u>
- Higher CR values for shear dominated laminates even from early loads → can indicate the dominant stress component and the consequent deterioration





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