

# Single-component hybrid simulation techniques for validation of fatigue models

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**Abstract:** Digital Image Correlation (DIC) is used to track the deformation of a cantilever beam at a measurement-point located away from the loading-point. A baseline test is run using the assumption of a linear relationship between the measurement point and the loading point. A second test is run that introduces a PID control based on the DIC measurements. This second method showed an improved ability to follow a cyclic command signal, with the X displacement improving from 14.1% to 6.1% error, the Y displacement from 3.8% to 1.25%, and the Z rotation from 3.2% to 2.0%.

**Keywords:** Digital Image Correlation; Hybrid Simulation; Displacement Tracking

## 1. Introduction

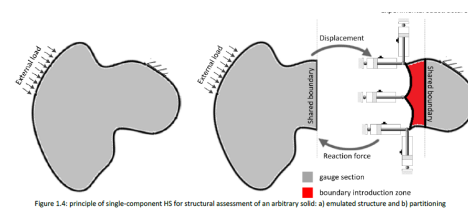
Hybrid Simulation (HS) is a sub-structuring technique that connects physical experiments with numerical analyses. Using HS for testing sub-structures reduces the size of the experimental set-up compared to full-scale testing by numerically modeling less critical sections. Parts of the structure that are critical to performance or behave in an unpredictable manner are tested in a physical substructure and then linked to the numerical model. [1]

To utilize the advantages of HS, the connection between the numerical and experimental parts must have minimal error. The reaction forces of the experimental substructure are returned to the numerical model for use in the next iteration. Errors in displacement result in errors in these reaction forces, which will cause the experimental substructure to be modeled as under- or over-stiff in the numerical model. Consequentially, accuracy of the applied displacements is paramount for successful HS.

### 1.1. Shared Boundary

The link between the experimental substructure and the numerical model is called the shared boundary (SB). As illustrated in Figure 1, the shared boundary is the interface where numerically calculated displacements are exported to the physical test software. Then load is applied to the experimental sub-structure until the specified displacements are achieved, obtaining a state of equilibrium on the shared boundary.

Though some applications allow for loading directly at the shared boundary, this test set-up requires a *boundary introduction zone* to prevent stress concentrations at the loading point from influencing the sub-structure. Because of this, the displacements of the shared boundary will differ from the displacements of the



**Figure 1.** Illustration of the Shared Boundary used in Hybrid Simulation[2]

32 actuators applying the load. This creates a challenge of obtaining the correct displacements at the shared  
33 boundary. This paper paper quantifies the advantages of using Digital Image Correlation (DIC) to track  
34 the displacement of the shared boundary.

## 35 2. Methods

### 36 2.1. DTU Hybrid Simulation Test Rig

37 The DTU Structural Laboratory contains a dedicated test rig for Hybrid Simulation. This HS Rig  
38 currently has one vertical actuator and two horizontal actuators attached to the free end of a cantilever  
39 beam made of pultruded fiberglass. The vertical actuator is an MTS 244.12, with a 25 kN force rating and  
40 an MTS 661.19F-08 25 kN load cell attached. The horizontal actuators are MTS 242.01, with a 4.5 force  
41 rating capacity and 661.19F-01 5 kN load cells attached to each.

42 A coupling matrix in the MTS 793.15 software relates the individual movements of the three  
43 actuators to an MDOF control at the loading point. This allows the loading point to be commanded in  
44 terms of three degrees of freedom: X displacement, Y displacement, and Z rotation. The dimensions of  
45 the rig and the capacity of the actuators are shown by the schematic in Figure 2.

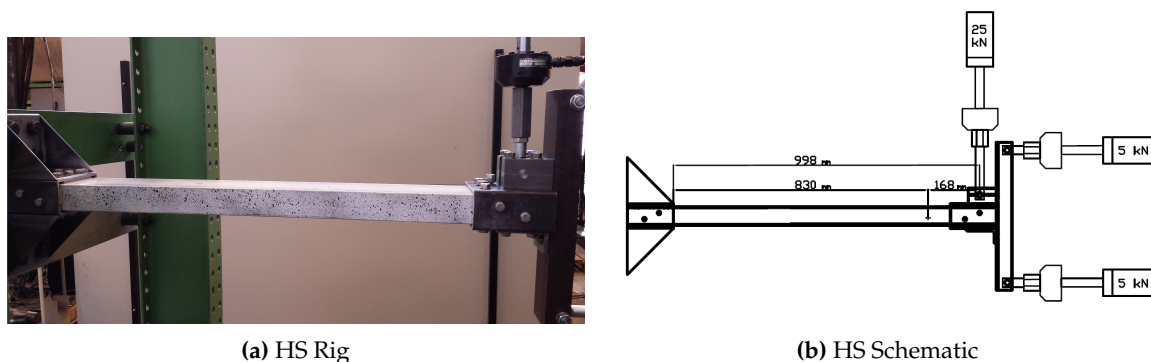


Figure 2. Hybrid Simulation Rig with a pultruded fiberglass cantilever beam

### 46 2.2. DIC System Set-up

47 A stereo-camera Digital Image Correlation system  
48 is set-up as part of the Hybrid Simulation Rig. The DIC  
49 system is a 12 megapixel system from GOM that uses  
50 Aramis software. A pattern of 7 tracking dots is applied  
51 to the side of the cantilever beam, shown in Figure 3.  
52 With a maximum sampling frequency of 232 Hz and a  
53 13ms time delay, the DIC system tracks the movement  
54 of these dots and calculates the average displacement in  
55 the X and Y direction and the rotation about the Z axis.  
56 The displacements can be measured to an accuracy of  
57 0.001 mm in the current configuration. The out of plane  
58 displacement can also be monitored, but is neglected in these tests because of the loading configuration.

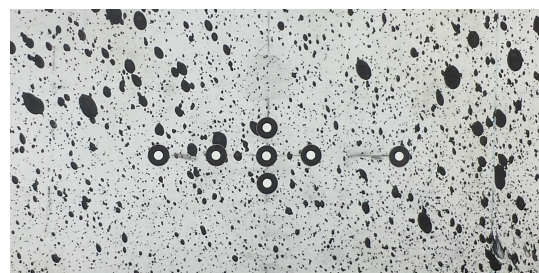


Figure 3. Pattern of DIC points for tracking displacement and rotation. The speckle pattern is unused.

59 The real-time measurements of these three degrees of freedom are sent via analog output to the  
60 MTS controller. The MTS software controlling the actuators is then able to access the DIC measurements,  
61 writing them to the output file or using them as part of the control loop.

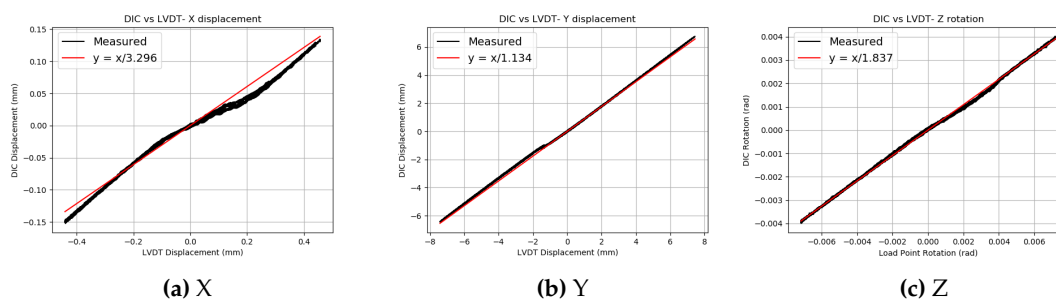
62 2.3. Measuring Experimental System Response

63 Because the experimental and numerical parts of HS connect through the shared boundary, the  
 64 displacements of the SB need to be monitored and controlled. The displacements and rotations at  
 65 the shared boundary are calculated by the MDOF coupling assuming a rigid connection between the  
 66 actuators and shared boundary point. However, in reality this is not a perfectly rigid connection, so the  
 67 relations between the nominal MDOF displacements and the actual SB displacement must be defined.

68 To determine these relations, loading patterns were applied for X displacement, Y displacement,  
 69 and Z rotation according to Table 1. The results are plotted in Figure 4 with a linear best fit line. The  
 70 inverse of these slopes is defined as a linear transfer function (LTF) that can be used to relate the desired  
 71 displacement or rotation to the required command signal. These tests isolate the three single DOFs in the  
 72 MDOF system, so no active interaction is observed.

**Table 1.** The linear transfer function (LTF) between the command signal and shared boundary displacement for 3 degrees of freedom

Test	DOF	Cycles	Load-point Amplitude	LTF
1	Y Displacement	30	$\pm 7.5$ mm	3.2962
2	X Displacement	30	$\pm 0.5$ mm	1.1336
3	Z Rotation.	30	$\pm 0.0075$ rad	1.8373



**Figure 4.** Best fit linear relationships between loading point and shared boundary point

73 2.4. Cyclic loading tests

74 A series of six tests was run to compare two different control methods. The tests used a cyclic  
 75 command signal for SB displacement.

76 Control Method 1 (CM1) multiplies the command signal by the linear transfer function (shown  
 77 in Table 1) to use as input to the MDOF coupling matrix. The coupling matrix converts the Shared  
 78 Boundary DOFs ( $d_x, d_y, \theta_z$ ) into displacements commands for the three actuators. For each actuator, an  
 79 inner PID loop is used to control the shaft displacement based on reading from an internal LVDT. No  
 80 feedback from the DIC system was used by CM1.

81 Control Method 2 (CM2) uses the same control loop as CM1, but includes an outer PID loop that  
 82 uses a feedback signal from the DIC system. The output of this outer PID loop is summed with the  
 83 output of the linear transfer function. These two control loops are visualized in Figure 5.

84 While the experimental rig is set up to run three degrees of freedom simultaneously, this experiment  
 85 looks to set the baseline by quantifying single degree of freedom (SDOF) performance. The experiments  
 86 are run at a frequency of 0.1 Hz, so their reaction is considered quasi-static.

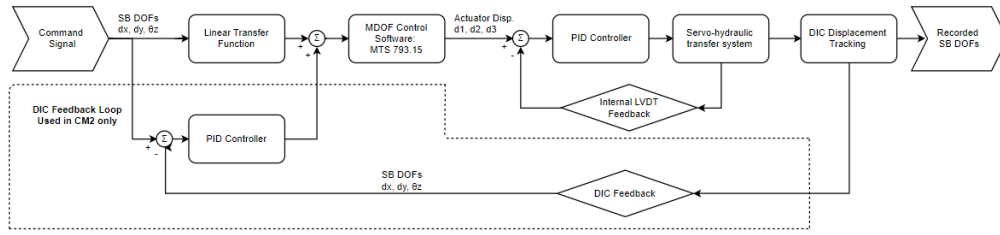


Figure 5. Control Loops for control methods CM1 and CM2

87 **3. Results**

88 The results of the tests are plotted in Figure 6 and the error functions are plotted in Figure 7. A  
 89 summary of the error values is presented in Table 2. It is shown in Figure 7 that the CM2 method starts  
 90 with a high error that reduces and stabilizes after 1 or 2 cycles. This is caused by error accumulating  
 91 in the integral portion of the PID loop before the test begins. This issue should be resolved for future  
 92 testing, but will be dealt with here by neglecting the first 10% of the test time and evaluating the error  
 93 only in the steady state region.

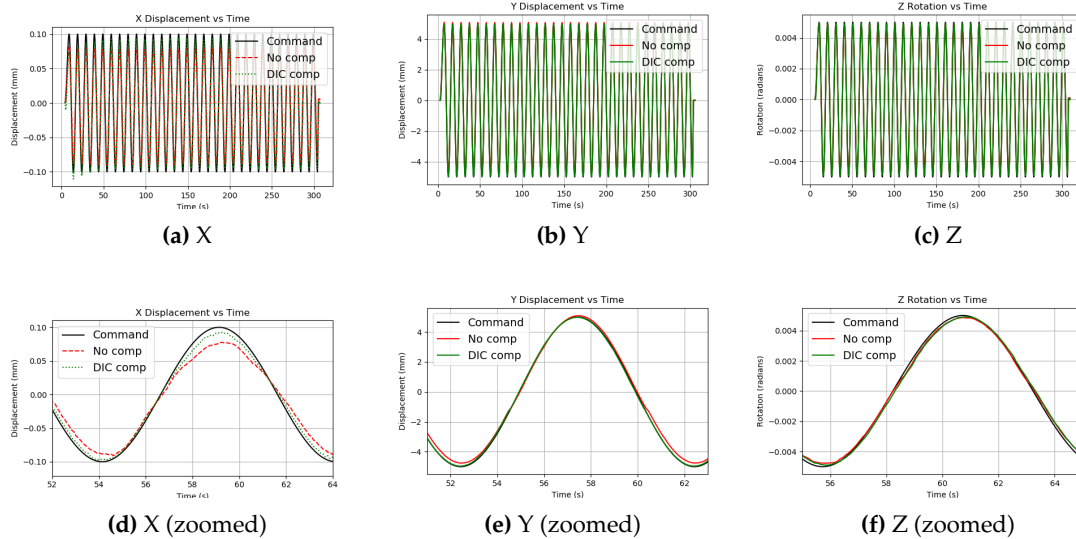
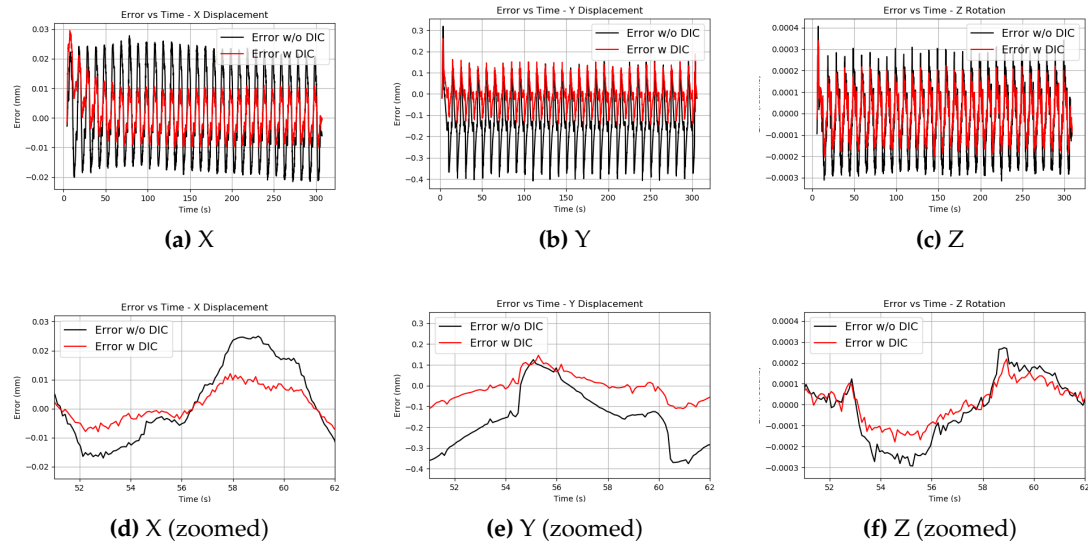


Figure 6. Shared Boundary Displacement/Rotation using two control methods



**Figure 7.** Error between the Shared Boundary Displacement/Rotation and the command signal

**Table 2.** RMS Values of Normalized Error (Steady State Region)

DOF	CM1	CM2
X-Displacement	14.1%	6.09%
Y-Displacement	3.78%	1.25%
Z-Rotation	3.23%	1.96%

## 94 4. Discussion

### 95 4.1. Observations on Experiments

96 The results in Table 2 show that the greatest error occurs in the X displacement. This is expected  
 97 since this DOF exhibits the least linear behavior of the three DOFs in Figure 4. It is suspected that worn  
 98 components in the horizontal actuators contribute to this non-linearity. Consequentially, the greatest  
 99 improvement from the introduction of DIC feedback with CM2 is on the X displacement. The more  
 100 linear responses of the Y displacement and Z rotation are also improved when using CM2, though to a  
 101 lesser extent.

### 102 4.2. Effects of Displacement Errors

103 An accurate application of loads is especially important in HS, where the reaction forces of the  
 104 experimental substructure are used to calculate a stiffness value applied to the numerical model. Errors  
 105 in displacement will cause the experimental substructure to be modeled as under- or over-stiff in the  
 106 numerical model. Consequentially, the next iteration of numerically calculated loads will be distorted by  
 107 the artificial stiffness change.

108 Errors in loading have additional consequences for fatigue tests. An error in displacement  
 109 corresponds to an error in the developed stresses and strains in the beam. Because of the logarithmic  
 110 nature of the S-N curve, a loading error can cause the fatigue life to be mispredicted by an order of  
 111 magnitude.

### 112 4.3. Future Work

113 While there is clear improvement from the use of CM2, an RMS error value over 6% in a quasi-static  
114 test is higher than what is desired. The planned fatigue tests will be run at a much higher frequency than  
115 0.1 Hz, so any control errors will likely be exacerbated the higher frequencies. Non-linearities exist in  
116 running the SDOF tests, so it is expected that MDOF tests will exhibit even more non-linear behavior.  
117 This makes accurate displacement control even more important.

118 Improvements will be made on the control system test equipment to reduce the error. The addition  
119 of lag compensation, peak-to-peak compensation, improved PID loop tuning, and filtering of the DIC  
120 feedback signal may also improve performance. Physical work will also be done to increase the linearity  
121 of the load train. Modifying the test rig and replacing worn actuators should decrease the non-linearity  
122 observed in the X displacement.

### 123 5. Conclusion

124 Digital Image Correlation was used to track the deformation of a cantilever beam at the shared  
125 boundary point. During quasi-static cyclic loading, the introduction of DIC feedback through Control  
126 Method 2 reduced the error in comparison to Control Method 1, with the X displacement improving  
127 from 14.1% to 6.1% error, the Y displacement from 3.8% to 1.25%, and the Z rotation from 3.2% to  
128 2.0%. The ability of DIC feedback to correct for the non-linear behavior of the experimental set-up  
129 provides justification for its inclusion in Hybrid Simulation. It is expected that system tuning, additional  
130 compensation, and hardware improvement can reduce the error even further.

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133 acknowledged.

### 134 Abbreviations

135 The following abbreviations are used in this manuscript:

136	DIC	Digital Image Correlation
	PID	Proportional-Integral-Derivative Controller
	HS	Hybrid Simulation
	MDOF	Multiple Degree of Freedom
137	SDOF	Single Degree of Freedom
	SB	Shared Boundary
	LTF	Linear Transfer Function
	CM	Control Method (1&2)

138

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