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# Analysis of Radiation Propagation inside a Hierarchical Solar Volumetric Absorber <sup>+</sup>

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Abstract: The Solar Receiver is a critical component of concentrated solar power technology; it works as a heat exchanger, transforming the concentrated solar radiation into high-temperature heat. Volumetric receiver technology, using air as heat transfer fluid, are designed to reach higher temperatures than the current receiver technology, which is limited by material resistance and fluid instability. The higher temperature, up to 1200 K, could be used in high-temperature industrial processes or a high-temperature thermodynamic cycle. A correct radiation propagation is essential to develop their performances, reducing reflection and emission losses and promoting the heat transfer to the fluid. In this study, the optical behaviour of Hierarchical Volumetric Receiver (HVR) developed in FBK has been studied using Monte-Carlo Ray Tracing (MCRT) simulations. The simulations have been validated in an experimental setup that evaluates the light transmissivity of the HVR porous structure. Two different HVR structures are evaluated with MCRT simulations that use a real solar dish geometry to configure a complete CSP plant. Results show that frontal and rear losses are, respectively, 12% and 3% of the incoming concentrated radiation. Inside the HVR, 15% of the incoming power is propagated trough the lateral void spaces. Therefore, the power spreading avoids the overconcentration of the centre of the focalized area. The HVR optical behaviour has been investigated, showing an optical efficiency of 85%.

**Keywords:** concentrated solar power; radiation propagation; volumetric receivers; monte carlo ray tracing; high-temperature heat

## 1. Introduction

The international agreements, which have been stipulated to contrast the global warming phenomenon, are aiming to reduce the CO<sub>2</sub> emissions and limiting the global average temperature increase to 2 °C [1]. CO<sub>2</sub> emissions together with other greenhouse gas (GHG) emissions are widely recognized as the main driver of global warming [2]. The heat production for the industrial sector and for the electricity generation represents more than 40% of the global GHG emissions[2]. Therefore, the research in carbon-free heat production need to be furtherly pushed. Among renewable energy sources, solar energy has the potential to alleviate the aforementioned energy issues [3]. Concentrated solar power (CSP) technologies use mirrors to concentrate the solar radiation and produce heat. CSP is considered as promising and viable technology capable of replacing fossil fuel consumption for the heat and power production[4]. Point focusing configurations, like central receiver towers or solar dish, are capable of higher concentration ratios than line focusing configuration [5]. High concentrated radiation, with fluxes over 500 kW/m<sup>2</sup>, can produce high temperature heat over 700 °C. This high temperature heat can be used to produce efficiently electrical

power or to feed high-temperature chemical processes (e.g., cement manufacturing, mineral and metallurgical extraction, solid waste gasification) [6], and advanced solar fuel synthesis [7].

A core component of these plants is solar receiver that is placed in the focal area of the concentrating mirrors. The receiver is essentially a heat exchanger which performs the conversion of concentrated radiation into heat. The state of the art of the point focusing CSP technologies is based on tubular solar receiver that can bear only moderate concentration ratios, using as Heat Transfer Fluid (HTF) oil, water-steam or molten salt. Therefore, the overall efficiency from solar to electricity is below 20% and the thermochemical routes are beyond reach [7]. According to the Carnot's theorem, higher temperatures in the heat source are essential to increase the thermodynamic power cycle efficiency, in CSP the heat source is represented by heat transfer fluid heated in the receiver [8]. The current tubular solar receiver technologies using molten salt as HTF are not suitable for higher concentrate flux and temperature (over 600 °C) due the salt instability and material mechanical resistance [9].

A promising path for the high temperature receivers is represented by the volumetric type [10]. These receivers use a porous structure that absorbs "volumetrically" the solar radiation [11]. In **Error! Reference source not found.** a sketch of receiver technologies working principles is reported. On the same side where the absorber is facing the radiation enters a gaseous HTF, typically air, which exchanges heat by convection cooling the absorber structure [12]. VR has some advantages if compared with TR. VR doesn't have any limitation in temperature coming from the HTF stability. They could achieve the so called "volumetric effect" that is described by Avila et al. [11] as "the temperature on the irradiated side of the absorber to be lower than the outlet temperature". This low temperature in front of the receiver reduces losses of radiative emission to the environment [13]. However only some VR have shown the presence of this effect in particular conditions [14].



Figure 1. Sketch of receiver working principles, on the left tubular receivers on the right volumetric technology.

VR receivers may have ordered or un-ordered structures that form their porous geometries. Both the categories could use ceramic or metallic materials. The first category may be monolithic [15–17], wire meshes [3], pinned [18]. The unstructured are mainly open-cell foams [13]. Some other authors have proposed packed bed configurations [19]. Hierarchical volumetric receiver (HVR) [20] was proposed to improve the performances of structured VR enhancing volumetric effect [14].

This paper presents a detailed analysis of the radiation propagation inside the hierarchical volumetric absorber using both experimental measurements and numerical simulations. In first part, the transmissivity of the H-VR prototype is estimated experimentally using as radiation source a solar simulator and a solar power meter as a probe. Considering this setup and the geometry of prototype the numerical simulations were conducted using the Monte Carlo Ray-tracing technique. The simulation has been validated comparing the simulation results with the experimental ones. In the second part taking into account the lab results, we modelled the performance of two HVR geometries coupled with a real concentrated solar system. In these simulations a solar dish geometry of a real prototype [21], Contest project, is used as concentrated source that hits the HVR geometry.

## 2. Methodology

In this section, the methodology applied in this study will be explained in detail: The methodology is thus divided in: (a) numerical part focused on the geometrical configuration and boundary conditions of MCRT simulations, (b) experimental part where is described the experimental setup designed to validate numerical simulations.

## 2.1. Numerical Methodology

Numerical simulations are conducted using the open-source Monte-Carlo ray tracing software Tonatiuh [22]. The HVR is simulated in two different simulation setups. In the first, the radiative flux transmitted through the receiver volume simulating similar conditions to the experimental setup, configuration 2. In the second, the receiver is simulated for a real CSP system in two different sizes configuration 1 and configuration 2, see Figure 2. The configuration 1 have the same hierarchical structure of configuration 2 but the receiver frontal area is larger, to allow ad adequate distribution of solar power provided by large parabolic dish, see **Error! Reference source not found.** 



**Figure 2.** (a) Frontal view of hierarchal receiver geometries that are numerically simulated, the configuration 1 is denoted by dimension A and B, configuration 2 is underlined in red. (b) Isometric view of Configuration 2. (c) Isometric view of Configuration 1.

#### 2.1.1. MCRT Collimated Source Simulations

These simulations are designed to validate our MCRT methodology comparing these simulations with measurements from experimental setup. Collimated source reproduces the experimental conditions illustrated in paragraph 0. Here the receiver geometry represented in Figure 2. The simulation configuration consists in a collimated radiative source, the HVR geometry and two auxiliary planes. The radiation source is reproducing the light coming out the Abet Solar simulator [23], the Direct Normal Irradiation (DNI) is set to 700 W/m<sup>2</sup>. The circumsolar ratio of the source is set to 4.65 mrad and the angle  $\alpha$  between the radiative source and receiver main axis is 90° deg. The receiver reflectance is set 40% with a non-specular behaviour. The auxiliary planes are two square planar surfaces, with 10 mm edge, parallel to the frontal surface of the receiver and orthogonal to main axis, represented in **Error! Reference source not found.**a. These planes are acting as probe for the incoming radiation. However, the planes aren't changing the photons behaviour and trajectories. The first is positioned next the receiver front face, underlined in **Error! Reference source not found.**b. The second, instead, is just behind the rear face. This configuration is designed to investigate the average irradiance on these surfaces measured as number of incident photons times the photon power divided by the surface area.



**Figure 3.** (a) Configuration of collimated source MCRT simulations: radiative source (above), HVR absorber geometry (center), auxiliary planes (below). (b) Geometrical arrangement in collimated source configuration underlining the angle between radiation source and receiver main axis.

#### 2.1.2. MCRT Simulations Using Concentrated Radiative Source

The HVR optical analysis is conducted on two receiver geometries. In the first a wider system is simulated, representing the absorber cup, in the second only the central cell, underlined in red in Figure 2, is under investigation. The bigger hexagon (Configuration 1), which represents the receiver frontal area, have the side that is the twice the smaller (Configuration 2). The results of complete cup simulation have a higher accuracy to represent the real behaviour of the radiation evolution that happen inside the HVR geometry. Indeed, in the first configuration several cells are simulated, thus is possible to evaluate the radiation evolution through non-solid lateral surfaces of HVR elementary cells. It has to be noted that this lateral radiation evolution has a great importance in the radiation propagation in the whole receiver. This importance is due to the capability of non-solid surfaces to spread the concentrated radiation. A correct radiation flux spreading through lateral surface is crucial

to avoid fluxes that are unbearable for the absorber structure. In addition to that, in the configuration 2 this evolution needs to be carefully evaluated to have a realistic solution, tuning properly the boundary conditions that are representing other parts of the whole absorber geometry (Cofiguration1). However, in the configuration 1 the computational effort is high, including geometry generation, simulation and post processing, needing more than two days to be completed.

Both receivers analysed in MCRT simulations uses a geometrical configuration that consists in: a solar source, an optical concentrator, hierarchical receiver and auxiliary planes.

The solar source, represented in Figure 4 by a black box, produces up to 10<sup>8</sup> photons with a pillbox sun-shape. The source has equivalent solar DNI of 0.001 W/mm<sup>2</sup> and circumsolar ratio of 4.65 mrad.

The optical concentrator is designed on geometrical generatrix of solar dish of the CONTEST project [21]. The solar dish is spherical-parabolical optical concentrator with an aperture diameter of 8.6 m. In central part of the dish there are spherical curved mirrors with a curvature of 9.293 m. The peripheral ring has, instead, a parabolic curvature, with a focal distance of 4.5 m. As parameters of the mirrors we set 85% of reflectivity and a specular error of 4 mrad.



**Figure 4.** MCRT simulation geometrical configuration, (**a**) Overall configuration: Contest Project solar dish (bottom), HVR receiver geometry (center) and Solar source (top). (**b**) Particular of the configuration: lateral auxiliary plane (right), receiver geometry (left grey), auxiliary section planes (left orange).

The HVR geometry is shown in Figure 4 on right part and is located at 4.5 m from the dish centre (i.e., in dish focal distance). Two types of planes are used as probe and as boundary conditions in MCRT simulations. The so-called auxiliary section planes, represented in orange in Figure 4, are parallel to the frontal face of the receiver. These planes are acting as virtual probes without any interaction with the photon path inside the receiver. Only the last plane, which is situated at the rear face, is a perfect absorber. The lateral auxiliary planes, represented in blue, are located in the lateral faces of the receiver, transversally to the absorber frontal face. These are intended to represent the insulating material around the receiver, for the configuration 1. On other hand, in small receiver simulation, the lateral planes are boundary conditions that are representing the transfer of photons between receiver cells. As aforementioned, due the concentrated radiation distribution, only a part of the radiative power that is coming from the central receiver cells to the lateral ones is coming back. The percentage of this power is evaluated using the absorber in configuration 1, hence in these simulations some additional auxiliary planes are added. These additional planes are located in the limit highlighted in red in Figure 2a.

## 2.2. Experimental Methodology

Experimental Measurements of Radiation Transmissivity

In this experimental part, the radiation transmitted through the whole HVR structure is measured. The experimental setup, represented in Figure 5 is formed by radiation source, the receiver prototype and radiation meter. The radiation source is the solar simulator "Sun 2000" by Abet technologies [23] equipped with a 1000 W xenon lamp and AM 1.5 filter. The solar simulator is capable of a maximum irradiance of 2000 W/m<sup>2</sup> that can be attenuated by a beam attenuator and power adjustment of the xenon lamp. The absorber prototype has the geometry reported in Figure 2b is kept in position by a plexiglass support. Just below the prototype is positioned the radiation meter model RS PRO Solar Power Meter ISM400 [24].



**Figure 5.** Experimental setup of radiation transmissivity through the HVR geometry: solar simulator (above), HVR prototype (center), radiation meter (below).

The irradiance measurement has been taken after two hours of solar simulator electronics power stabilization.

#### 3. Results

#### 3.1. Result of Transmissivity Experimental Setup

In **Error! Reference source not found.** are reported the experimental results of Irradiance through the HVR absorber under a collimated source. The input irradiance that comes from the solar simulator is reported in table first row. The reported value is the average of ten measurements with a value of 365 W/m<sup>2</sup> and a standard deviation of 2 W/m<sup>2</sup>. The light transmitted through the receiver geometry have been measured ten times in different zones of the receiver. As result the transmissivity has an average irradiance of 33.8 W/m<sup>2</sup> and a standard deviation of 5.6 W/m<sup>2</sup>. These transmittance

results point out that about a tenth (9.2%) of incoming radiative power is lost in the receiver rear face. This high level of transmitted power is due to characteristics of impinging radiation, similar to the solar irradiance, that has an aperture angle of only 4.65 mrad far lower than design aperture angle.

Zone	Irradiance W/m <sup>2</sup>	Transmissivity
Without Receiver + STDEV	$365 \pm 2$	-
Average + STDEV	$33.8 \pm 5.6$	$9.2\% \pm 1.6\%$

Table 1. Result of radiative transmissivity experimental setup.

#### 3.2. Result of Simulations

#### 3.2.1. Collimated Source Simulations

As described in the methodology paragraph, these simulations are reproducing the conditions in experimental setup. As reported in the first row of Error! Reference source not found., the simulation, with the ideal conditions, has average front irradiance of 367.2 W/m<sup>2</sup> and rear of 46 W/m<sup>2</sup>. As result the transmissivity is 12.5% that is almost 3% more of the experimental one. This overestimation of the transmissivity could be due to the perfect behaviour of the simulation if compared with experimental condition. Unfortunately, in the experimental setup a perfect perpendicular condition between Solar simulator and receiver main axis cannot be not guarantee. Moreover, the reflectivity 53% is an average value of spectral reflectivity reported in the literature on flat surface. The surface of materials is quite rough (Ra > 3 micron) this mean that a reduction of reflectivity can be observed [14]. Considering the ideal case and the worst case with reflectivity value of 30% and the non-perpendicular condition average irradiances in the rear face has been calculated. Therefore, decreasing  $\alpha$  of 3° Deg the average rear irradiance reduces to 33.6 W/m<sup>2</sup> and 32.5 W/m<sup>2</sup> for 53% and 30% reflectivity respectively. The transmissivity drops to 9.2% and 8.9%. The average value of transmissivity considering a reflectivity of 53% is  $11.3\% \pm 1.5\%$ , see Error! Reference source not **found.** While the average amount of transmissivity considering a reflectivity of 30% is  $10.9\% \pm 1.5\%$ . Therefore, the experimental transmissivity  $(9.2\% \pm 1.6\%)$  have a good overlap with simulated transmissivity. This result indicates that the simulation can produce robust results.

#### 3.2.2. Photon Number Sensitivity Analysis

In the considered MCRT simulations only a part the photons reach receiver geometries. Hence, the heat flux inside the receiver could depend on number of photons generated by the solar source, which are evolving then in the considered system. A sensitivity analysis of this flux inside the receiver is carried out with a generated photon number range between 25 and 200 million. In **Error! Reference source not found.** the evolution of radiative power along the Configuration 1 receiver main axis is reported and six different simulation results are compared. From the comparison, it appears clearly that the evolution of radiative power doesn't depend on photons number. The variation of incoming power at radiation that enters in receiver front face between 25 and 200 million simulation is less than 0.005%.

Therefore, the number of 100 million photons is chosen as compromise for standard number for simulations.

Source Conditions Angle to Receiver Main	Average Irradiance Front	Average Irr Surfac	radiance Rear e [W/m²]	Transmis	ssivity %
Axis [Deg]	Surface W/m <sup>2</sup>	ρ = 53%	ρ = 30%	ρ = 53%	ρ = 30%
90°	367.2	46	45.9	12.5%	12.5%
89°	365.9	44.2	43.4	12.1%	11.9%
88°	365.7	37.9	39.1	10.4%	10.7%

**Table 2.** MCRT simulations with collimated source varying the angle between radiation source and receiver main axis. Two different receiver reflectivity  $\rho$  are considered 53% and 30%.

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87°	364.6	33	32.5	9.2%	8.9%
Average	365.9	$41.5\pm5.7$	$40.2\pm5.8$	$11.3\% \pm 1.5\%$	$10.9\% \pm 1.6\%$

3.2.3. MCRT Simulation Results with Concentrated Source

In order to evaluate the evolution of radiation inside the HVR receiver a complete postprocessing routine was developed. This post-processing routine analyses the irradiance on every auxiliary plane, including all the lateral and cutting planes of the configuration 1.



**Figure 6.** Irradiance power evolution on forward direction varying the number of photons: six different simulation from 25 to 200 million of photons.

**Error! Reference source not found.** shows the relative radiative power evolution inside the HVR receiver in the Configuration 1. All results are weighted to incoming power from the Contest solar dish that is 1.6015·10<sup>4</sup> W. This incoming radiation evolves in the forward direction from the frontal face to the rear decreasing to less 2% of the incoming power that leaves the HVR rear face. It may be noted that the forward power trend has four exponential decays. These decays depend on the different HVR layers that radiative power is hitting on the way to rear face. The exponential decay gets slower in the last part of every layer due to lower presence of the solid body. As well as on forward direction, the evolution of backward power is influenced by solid geometry. This effect could be observed in the reversed decay of the backward direction power, especially from 15 mm to 10 mm where the second layer is absorbing or reflecting more than 5% of the total incoming power.



**Figure 7.** Relative Radiative power evolution along main axis of Configuration 1: Radiative power moving on forward direction (from front to rear absorber face), Backward direction, Lateral non-absorbed power, Absorbed power.

The power absorption also shows a trend that is influenced by the presence of solid material. On this parameter it should be noted the importance of the frontal surface of third layer, at 15mm from the frontal surface, that absorbs slightly less than 20% of the incoming power.

The power spread on lateral non-solid surfaces through the whole absorber is 5.3% of the total incident power. Most of this power is dispersed in the first and second layer of absorber geometry. Only 0.1% of radiation spread on lateral non-solid surface, therefore the first part of absorber makes the ongoing flux more collimated.

**Error!** Reference source not found. is reporting a comparison of the power evolution along the main axis of the two configurations. Three different solutions are compared: Configuration 1, Configuration 2 and Configuration 1 central. The Configuration 1 central utilizes the results of simulations of Configuration 1 but only the volume of the configuration 2 is considered in postprocessing of the results. It may be noted that there are some important differences of forward radiation power between Configuration 1 and 2 in absorber first part. The configuration 1 power decays more rapidly than the other. At the end of layer 1 the configuration 1 power is 2% lower than the configuration 2. If this comparison is moved to configuration 1 central part only this difference reduces to only 1%. These differences are nullified in the receiver second part where decreases to 0.2%. The different behaviour between the two configurations could be due to boundary conditions of configuration 2. These conditions change the evolution of radiative power through non-solid lateral surfaces that, as explained earlier, have an important influence in the first and second layer. Although the differences in radiation evolution Configuration 2 performs well in receiver losses with a variation from Configuration 1 of 0.2% in both front and back losses. The configuration 2 has demonstrated a good overlap compared with Configuration 1. This means the optical performance of volumetric receivers with this hierarchical structure has as feature the scale invariance. We remark that this feature is very important to scale up at the industrial process HVR.



**Figure 8.** The relative radiative power attenuation along HVR main axis in both the considered configurations. The power is evaluated on the forward, from frontal face to the rear one, and backward direction.

## 4. Conclusions

In this work a detailed analysis of radiation propagation inside the hierarchical absorber was presented. This analysis includes both experimental measurements and numerical models. In the first part, using the experimental result, a model MCRT simulations with a radiative collimated source. In second part a MCRT simulations analyses HVR behaviour in two different configurations using a real concentrating mirror geometry. A comparison of results between the here developed experimental protocol and the MCRT simulation shows a good concordance indicating that the protocol is robust. The numerical simulation results have demonstrated a good concordance between the two HVR with different dimensions. This mean that the use of a reduced receiver geometry allows a time and computational effort reduction that enables new investigation prospects of several geometries and conditions. A new experimental setup that could measure the power spread and reflected in the receiver geometry, is under development. The authors remark that this methodology can be used for other receiver geometries.

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## Abbreviations

- CSP concentrated solar power
- HVR hierarchical volumetric receiver
- HTF heat transfer fluid

- MCRT monte carlo ray tracing
- STDEV standard deviation

## List of symbols

- $\alpha$  angle between receiver main axis and radiative source
- ρ reflectivity

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